



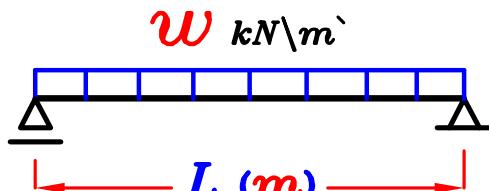
## **2<sup>nd</sup> ARCH-RC**

### **RC COURSE OUTLINES**

- 1. STRUTURE REVISION.**
- 2. DESIGN OF BEAMS FOR FLEXURE.**
- 3. DESIGN OF BEAMS FOR SHEAR.**
- 4. REINFORCEMENT DETAILS.**
- 5. DESIGN OF SHORT COLUMNS.**

سنحتاج في البداية أن نتذكر رسم الـ **B.M.D.** للكمرات

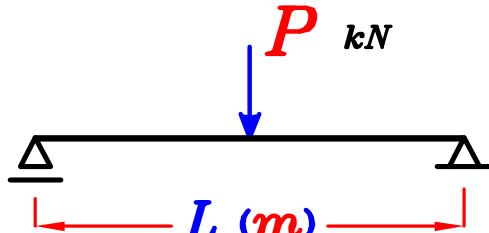
## ① Simple Beam subjected to distributed Load.



The bending moment diagram is a downward-opening parabola. The formula for the bending moment is:

$$M = \frac{wL^2}{8}$$

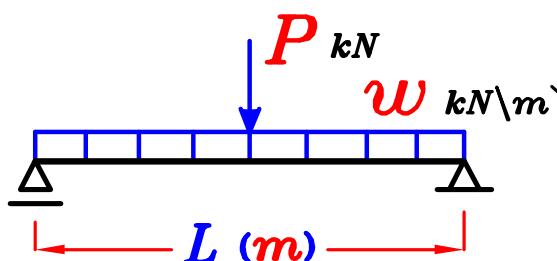
## ② Simple Beam subjected to concentrated Load at the mid. span.



The bending moment diagram is a triangle. The formula for the bending moment is:

$$M = \frac{PL}{4}$$

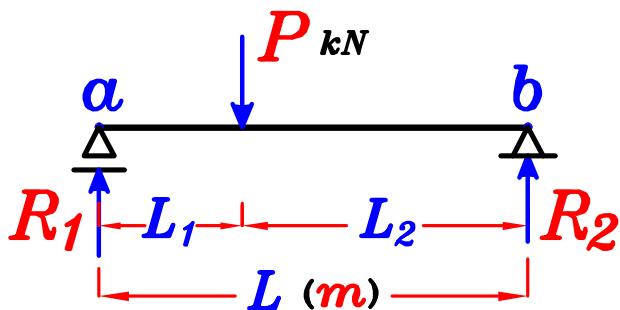
## ③ Simple Beam subjected to distributed Load + concentrated Load at the mid. span.



The bending moment diagram is a parabola. The formula for the bending moment is:

$$M = \frac{wL^2}{8} + \frac{PL}{4}$$

## ④ Simple Beam subjected to concentrated Load not at the mid. span.

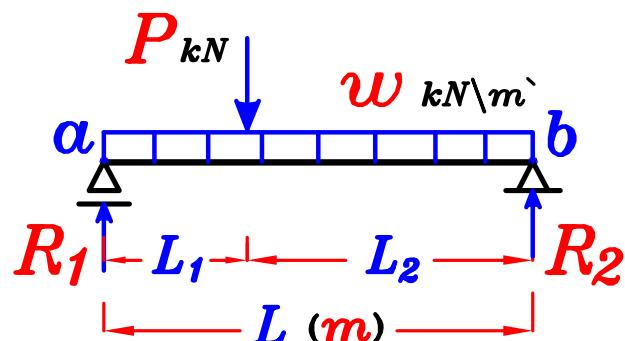


$M = R_1 * L_1 = R_2 * L_2$

$$\sum M_b = \text{Zero} \xrightarrow{\text{Get}} R_1$$

$$\sum Y_a = \text{Zero} \xrightarrow{\text{Get}} R_2$$

⑤ Simple Beam subjected to distributed Load + concentrated Load not at the mid. span.



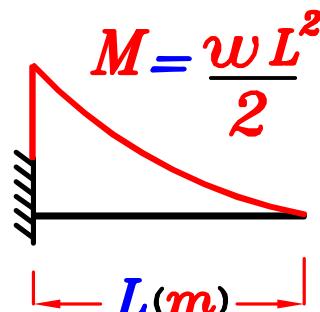
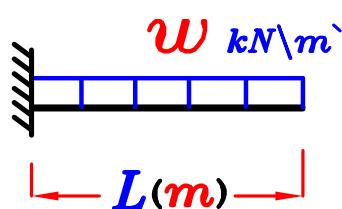
$$M = R_1 \cdot L_1 - \frac{w \cdot L_1^2}{2}$$

$$M = R_2 \cdot L_2 - \frac{w \cdot L_2^2}{2}$$

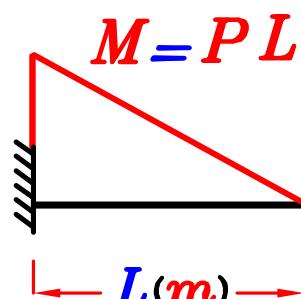
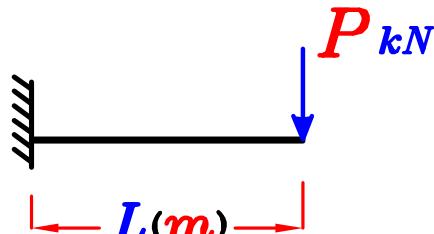
$$\sum M_b = \text{Zero} \xrightarrow{\text{Get}} R_1$$

$$\sum Y_a = \text{Zero} \xrightarrow{\text{Get}} R_2$$

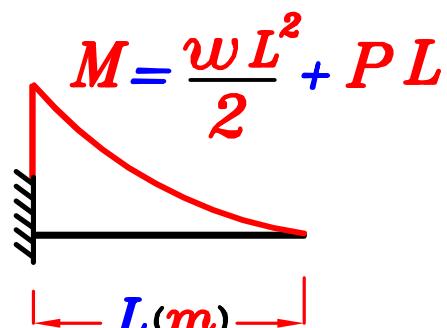
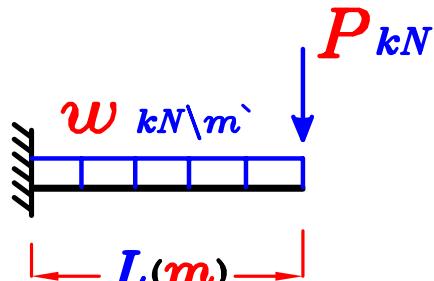
⑥ Cantilever subjected to distributed Load.



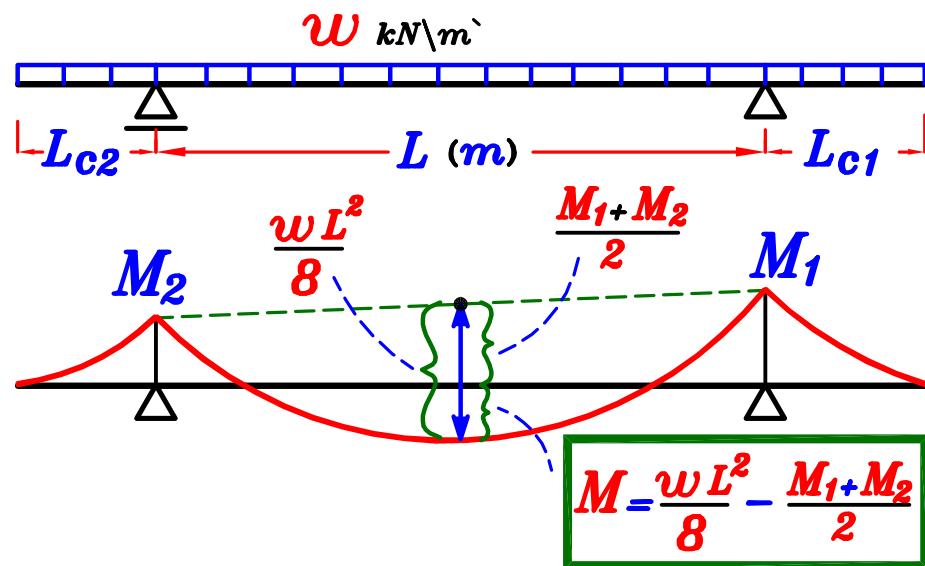
⑦ Cantilever subjected to concentrated Load at the End of the cantilever.



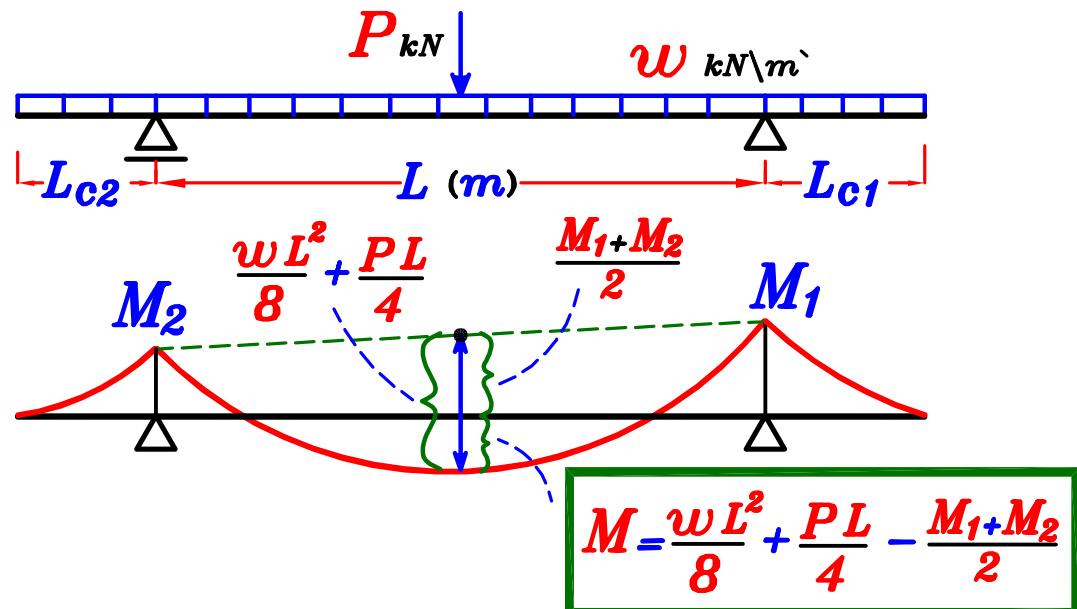
⑧ Cantilever subjected to distributed load + concentrated Load at the End of the cantilever.



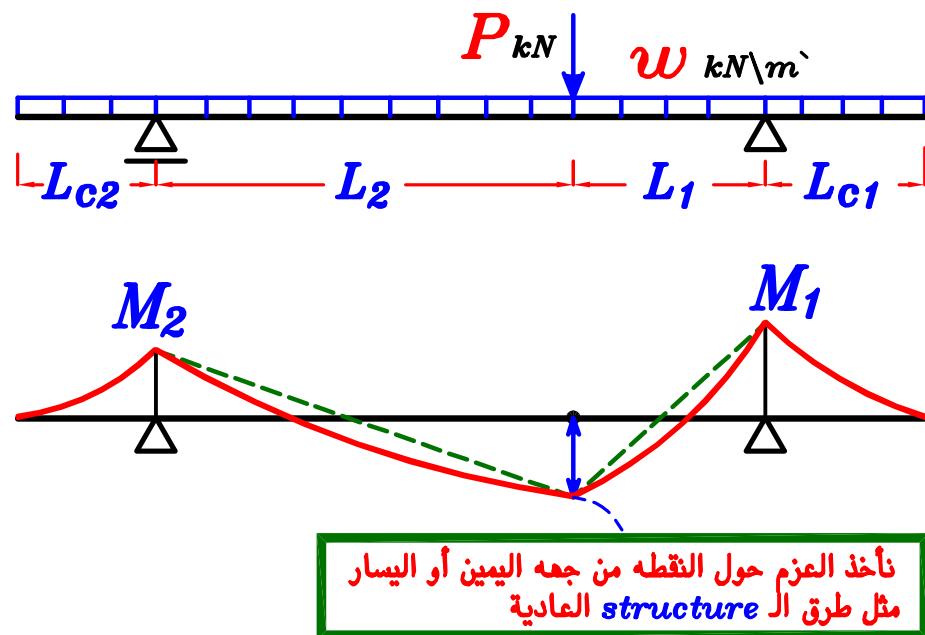
**(9) Beam with two cantilever subjected to distributed Load.**



**(10) Beam with two cantilever subjected to distributed Load + concentrated load at the mid. span.**



**(11) Beam with two cantilever subjected to distributed Load + concentrated load not at the mid. span.**



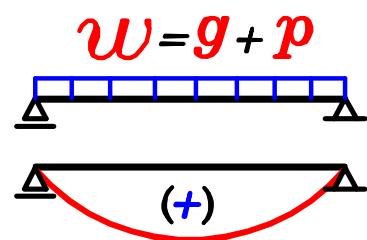
## Absolute (Max-Max) B.M.D. For Beams.

توجد لكل كمرة عده حالات من التحميل لتحديد أكبر عزوم على الكمرة .

### ① Simple Beam.

Max. (+ve) B.M.

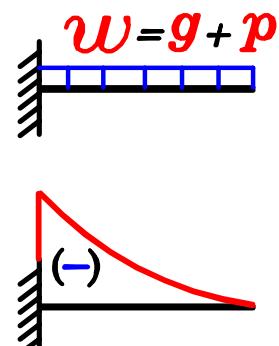
No (-ve) B.M.



### ② Cantilever Beam.

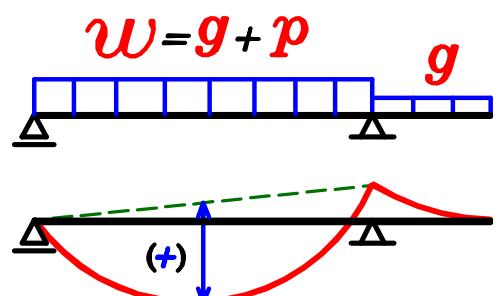
Max. (-ve) B.M.

No (+ve) B.M.

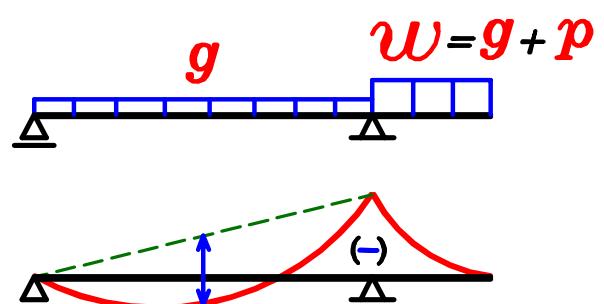


### ③ Beam with Cantilever.

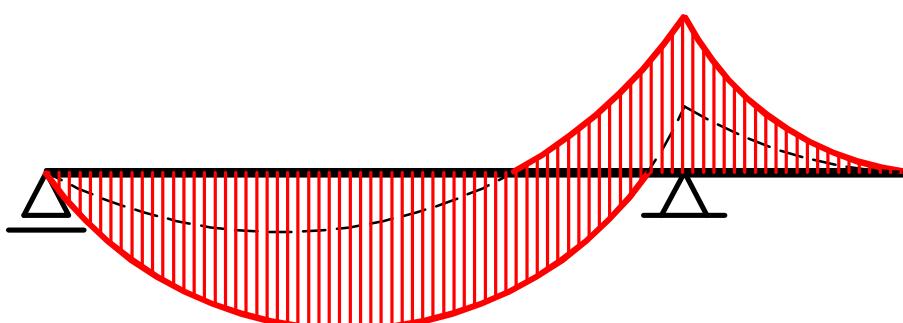
a) Max. (+ve) B.M.



b) Max. (-ve) B.M.

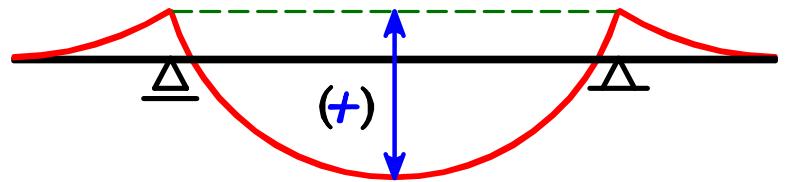
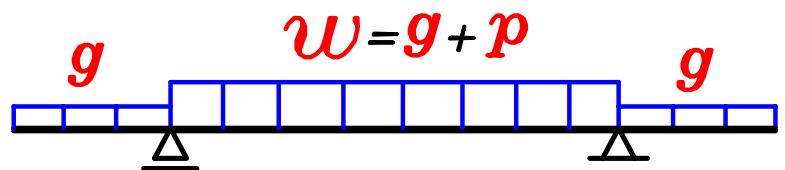


### Max-Max B.M.D.

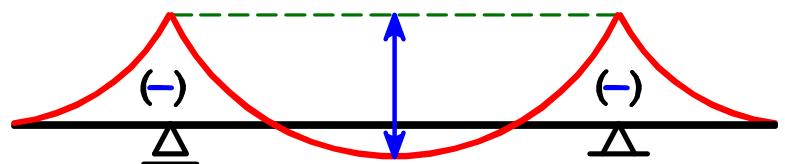
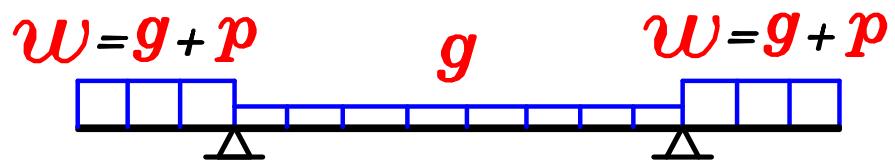


**④ Beam with 2 Cantilevers.**

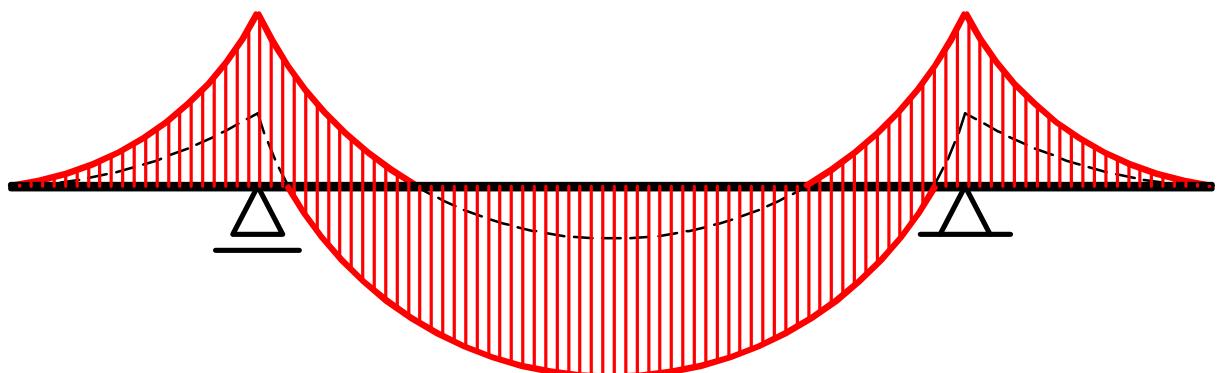
**a) Max. (+ve) B.M.**



**b) Max. (-ve) B.M.**



**Max-Max B.M.D.**



# Example.

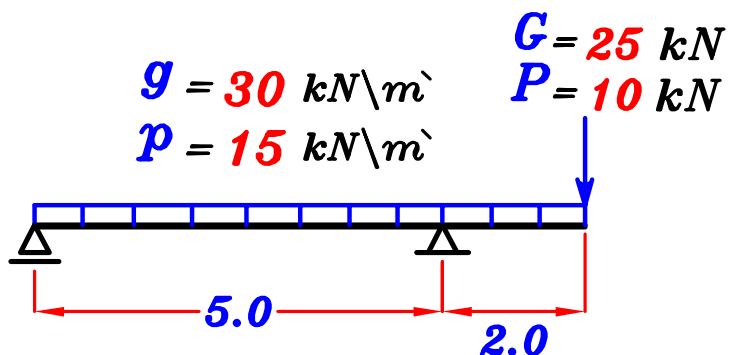
**Draw Max-Max B.M.D.**

$g$  = D.L. (Distributed Load)

$G$  = D.L. (Concentrated Load)

$p$  = L.L. (Distributed Load)

$P$  = L.L. (Concentrated Load)

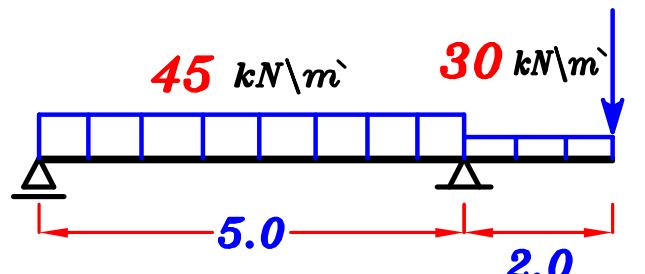
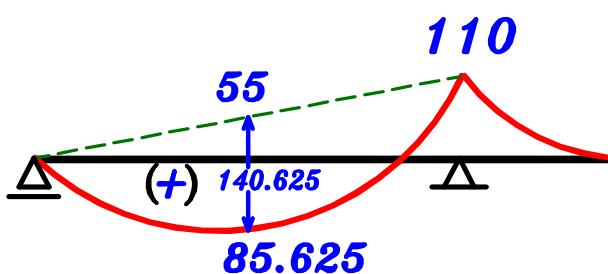


**Solution.**

a) Max. (+ve) B.M.



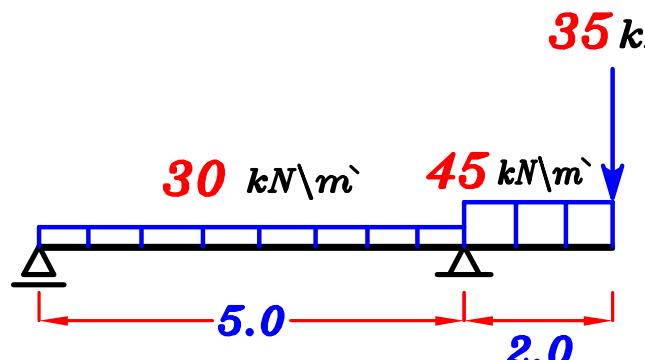
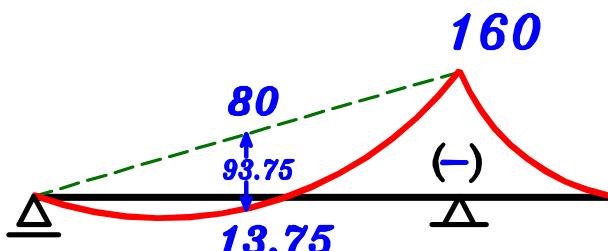
$25 \text{ kN}$



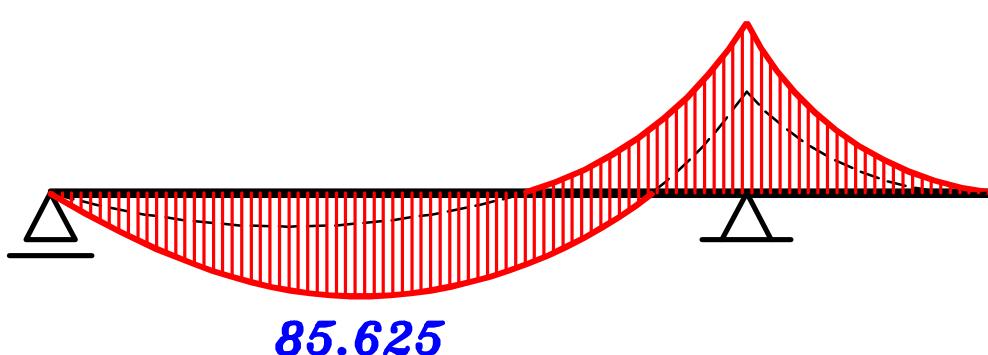
b) Max. (-ve) B.M.



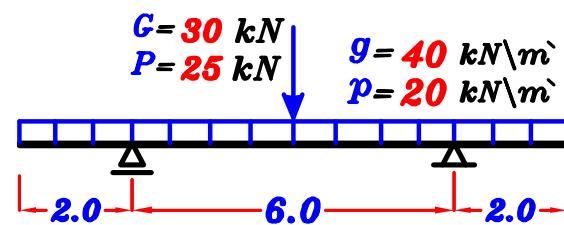
$35 \text{ kN}$



**Max-Max B.M.D.**



# Example.

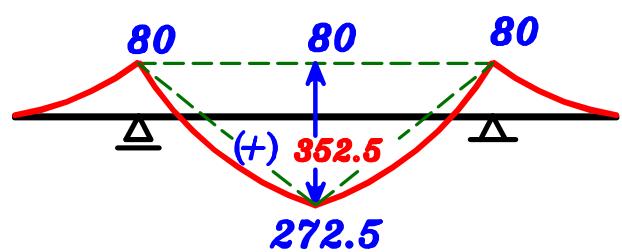
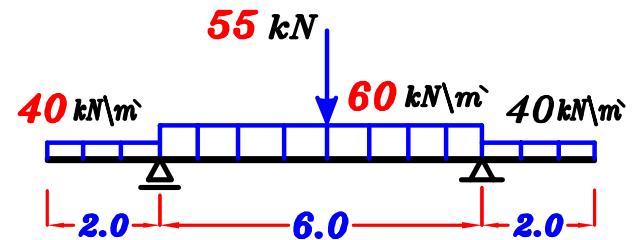


Draw Max-Max B.M.D.

@ Max. (+ve) B.M.



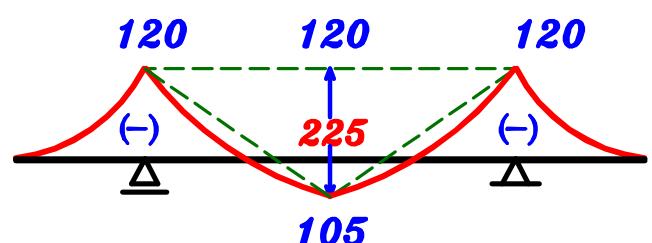
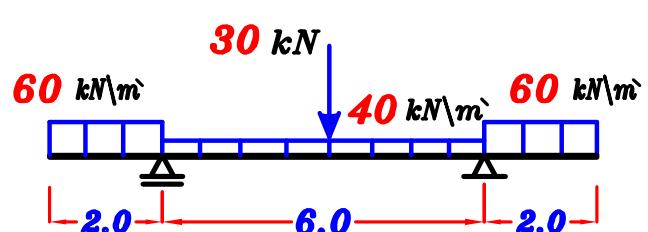
$$\frac{wL^2}{8} + \frac{PL}{4} = \frac{60(6)^2}{8} + \frac{55(6)}{4} = 270 + 82.5 = 352.5$$



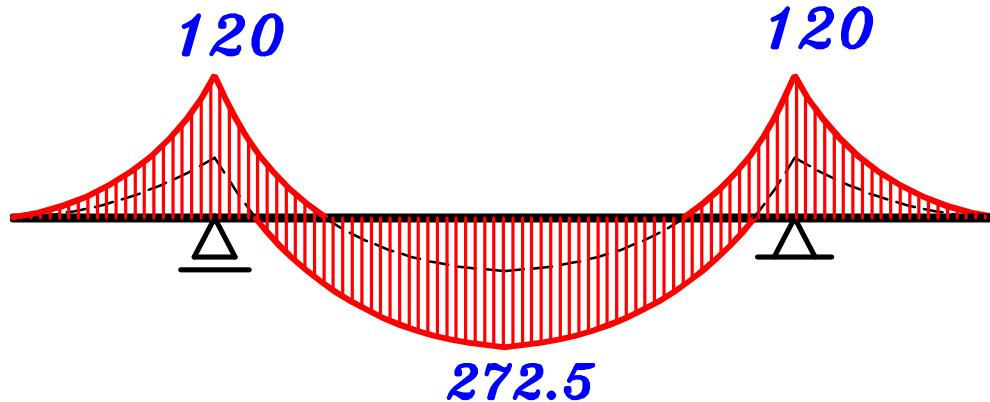
(b) Max. (-ve) B.M.



$$\frac{wL^2}{8} + \frac{PL}{4} = \frac{40(6)^2}{8} + \frac{30(6)}{4} = 180 + 45 = 225$$

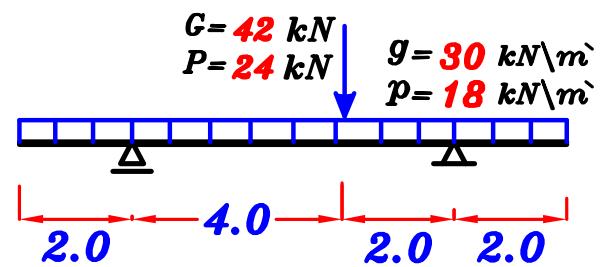


Max-Max B.M.D.

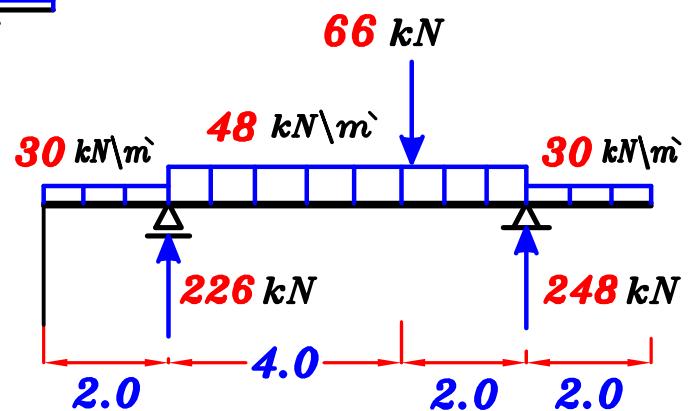
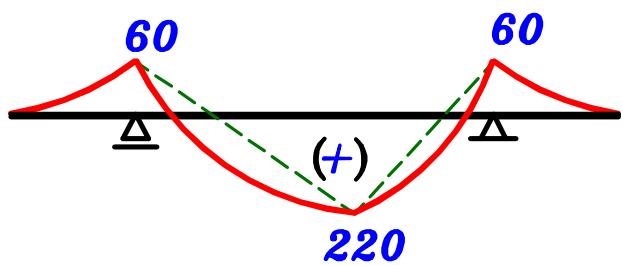


# Example.

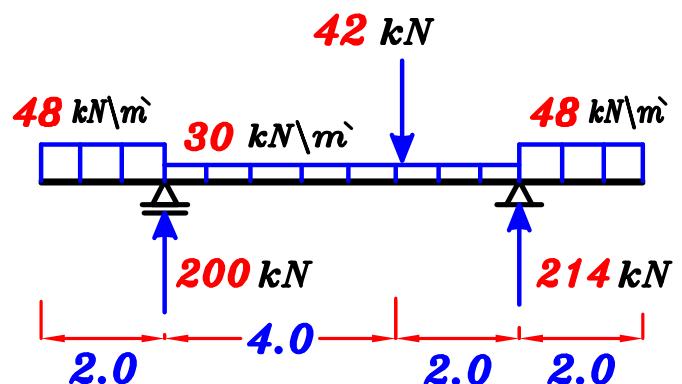
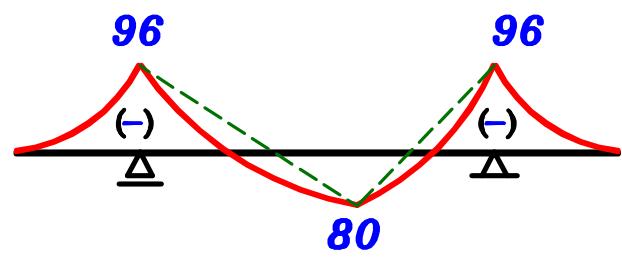
**Draw Max-Max B.M.D.**



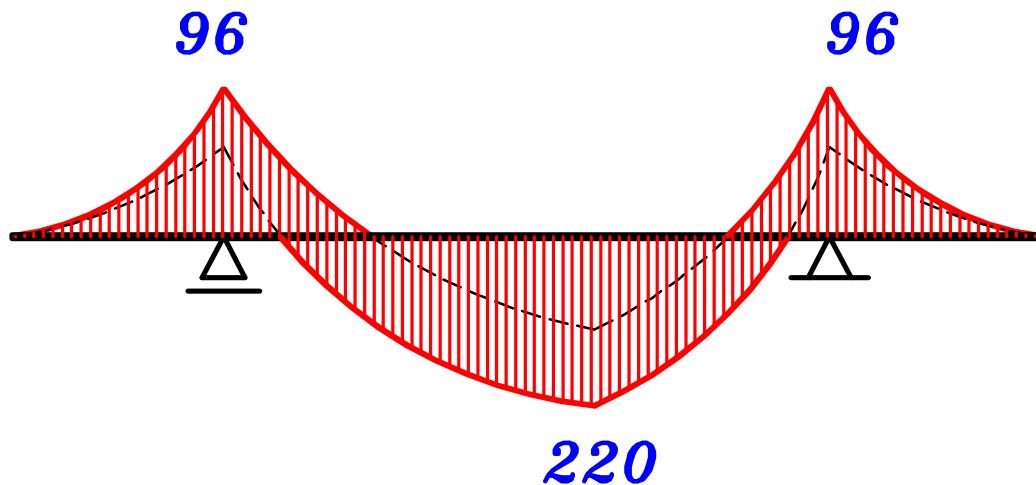
**@ Max. (+ve) B.M.**



**(b) Max. (-ve) B.M.**



**Max-Max B.M.D.**

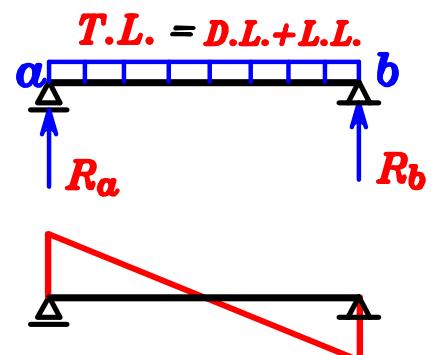


# **Max-Max S.F.D.**

ترتبط ال **Reactions Shear Force** بال **Reactions** لذلك عند رسم **Max Shear Force** نعمل على أن تكون ال **Reactions** أكبر قيمة لها.

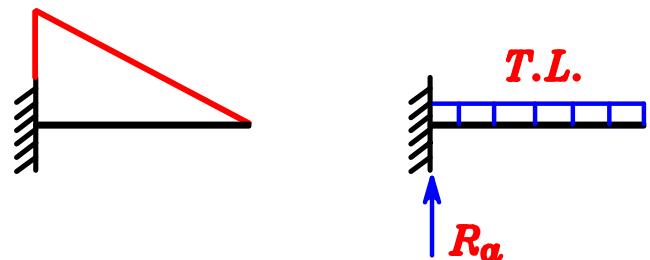
## ① Simple Beam.

$$\text{Max. } R_a, \text{ Max. } R_b$$



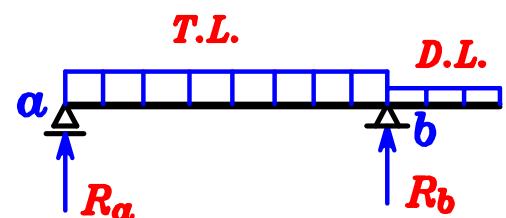
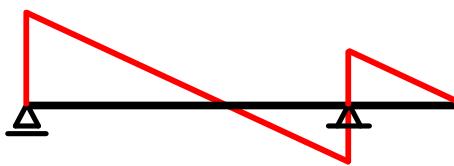
## ② Cantilever Beam.

$$\text{Max. } R_a$$

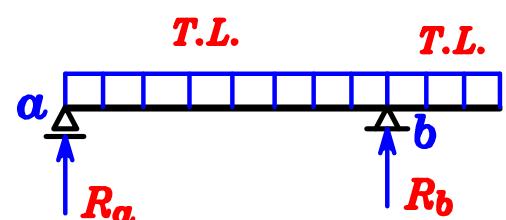
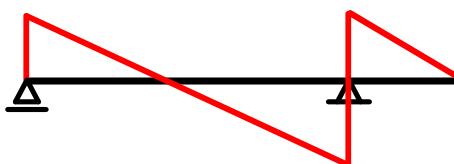


## ③ Beam with Cantilever.

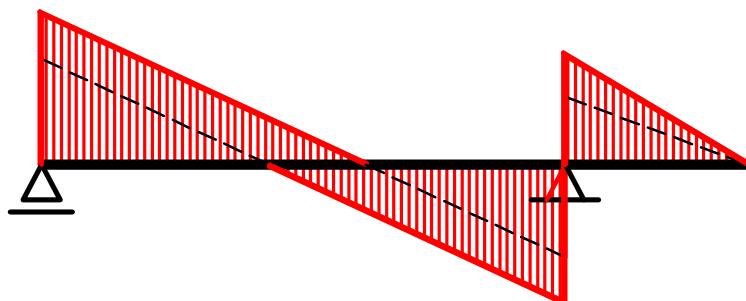
$$① \text{ Max. } R_a$$



$$② \text{ Max. } R_b$$

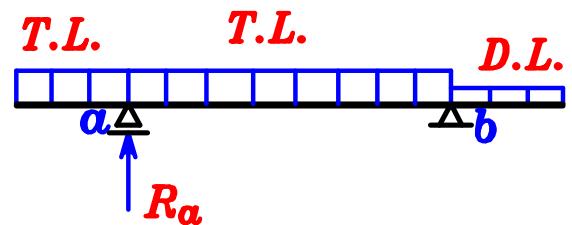
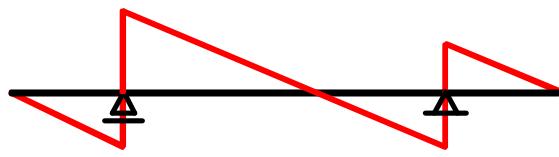


## Max-Max S.F.D.

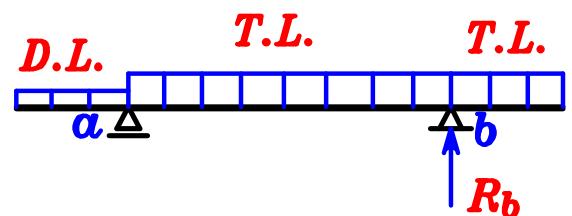
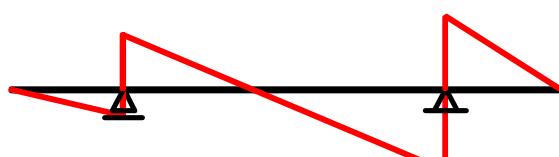


#### ④ Beam with 2 Cantilevers.

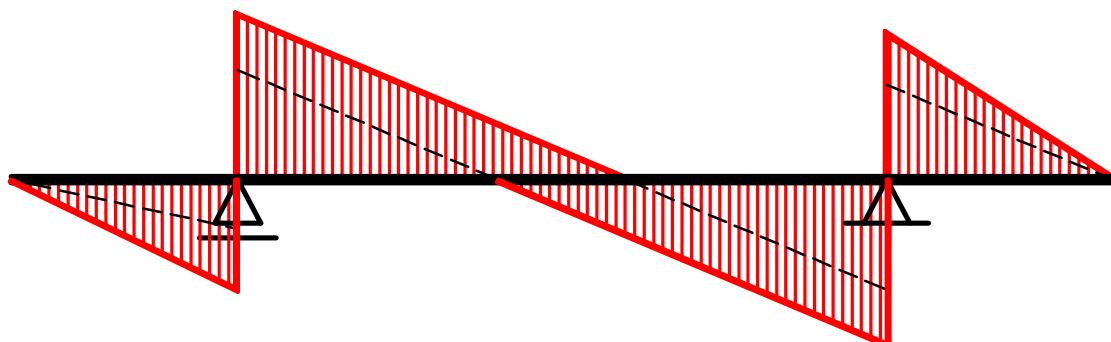
① Max.  $R_a$



② Max.  $R_b$



**Max-Max S.F.D.**



**ملحوظه هامة**

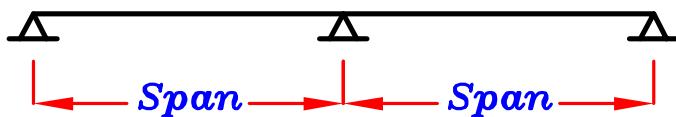
ممكن للتسهيل أخذ جميع حالات تحميل الـ **Shear** حاله واحده فقط و هي **T.L.** على كل الكمراه .  
وستكون النتائج قريبيه من الـ **max-max Shear** و بالطبع هذه طريقة تقريبيه .

# Continuos Beams.

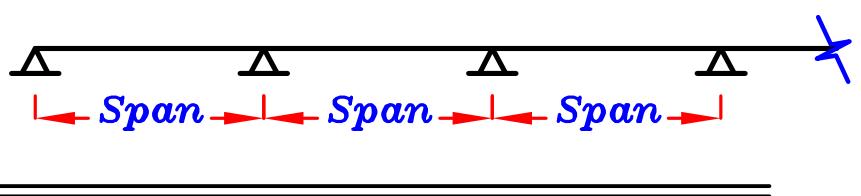
## الكمارات المستمرة .

الكمارات المستمرة هي كمارات لها أكثر من بحر و لها أكثر من ركائزتين **2 Supports**

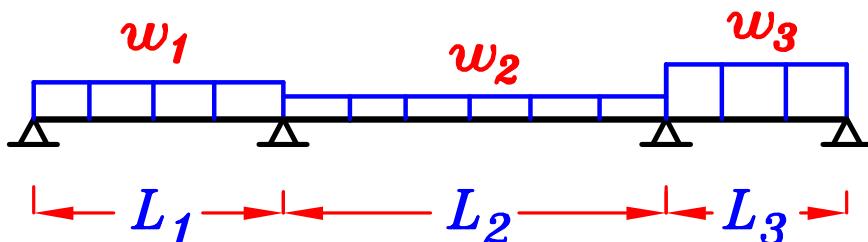
### ① 2 Spans.



### ② More than 2 Spans.



و هناك قيم ل **max-max S.F.D.** و **قيم ل max-max B.M.D.**



ممكن إستخدامها مباشرة بشرطين :

- ١ - أن تكون بحور الكمارات متساوية .
- أو أن يكون الفارق بين أكبر بحر و أقل بحر لا يزيد عن ٢٠ %

$$\frac{L_{\max} - L_{\min}}{L_{\min}} > 20 \%$$

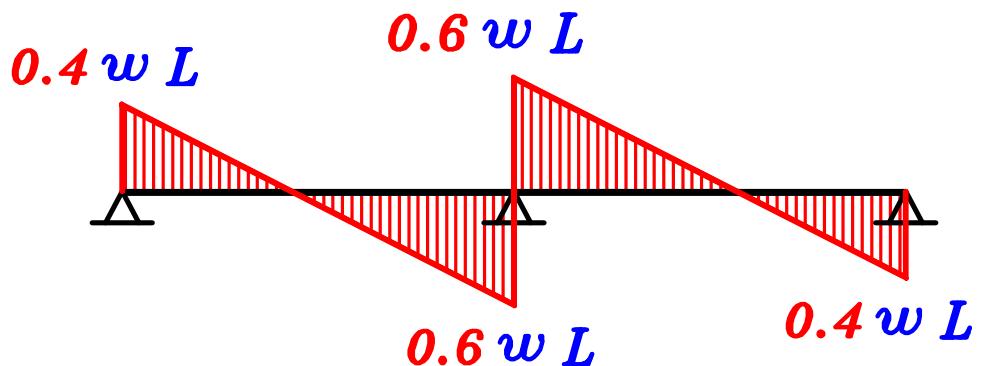
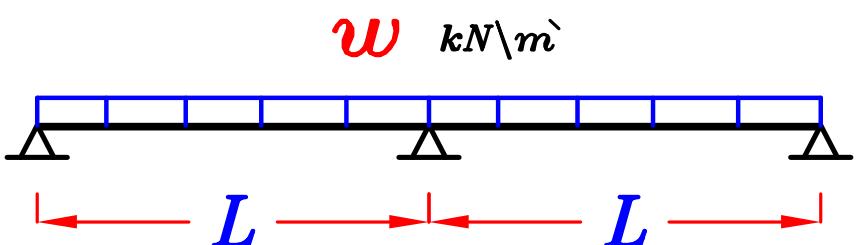
- ٢ - أن تكون الأحمال على البحور متساوية .

- أو أن يكون الفارق بين أكبر حمل و أقل حمل لا يزيد عن ٢٠ %

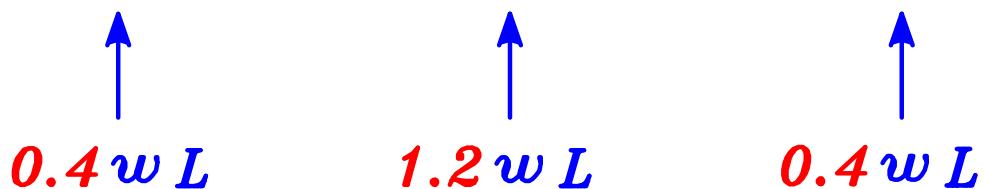
$$\frac{w_{\max} - w_{\min}}{w_{\min}} > 20 \%$$

# ① Continuous Beam with 2 spans.

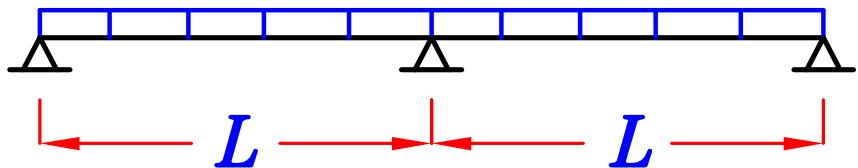
*Load For Shear*



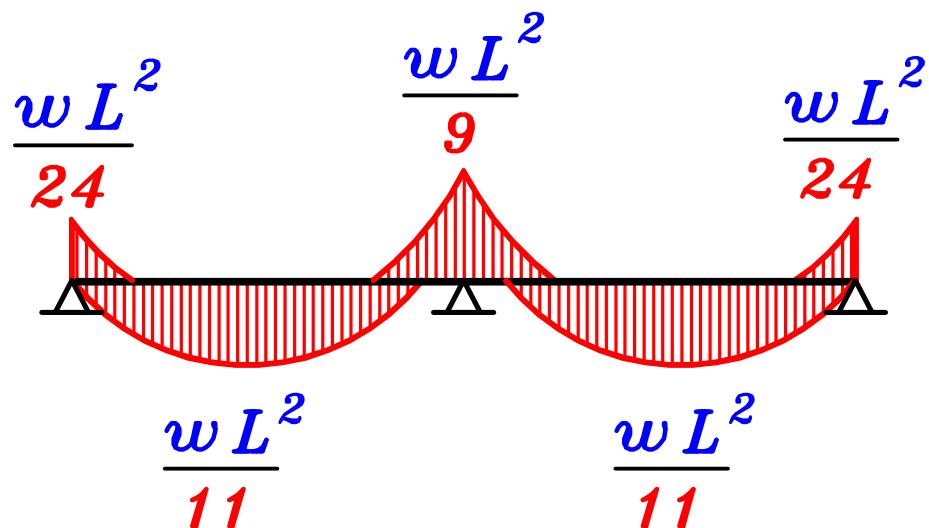
*Reactions.*



*Load For Moment*



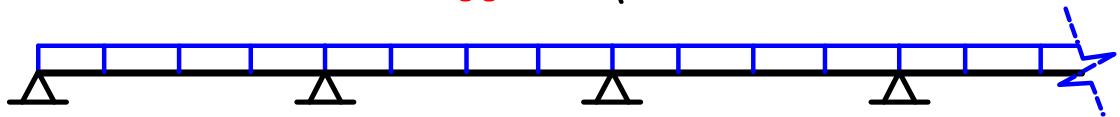
*B.M.D.*



## ② Continuos Beam with more than 2 spans.

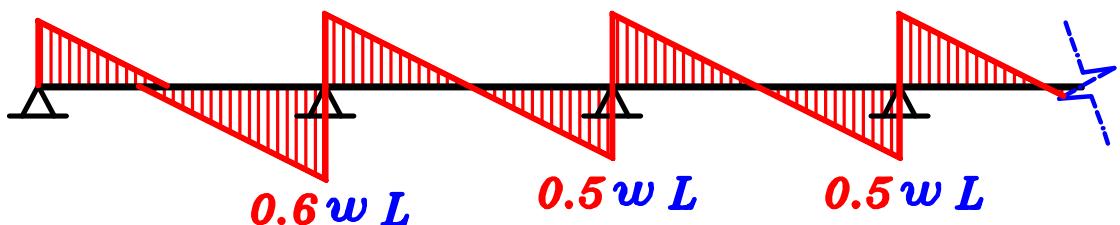
*Load For Shear*

$$w \text{ kN/m}$$



$$0.45wL \quad 0.5wL \quad 0.5wL \quad 0.5wL$$

*S.F.D.*

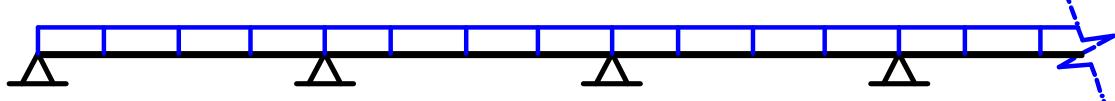


*Reactions.*

$$0.45wL \quad 1.1wL \quad 1.0wL \quad 1.0wL$$

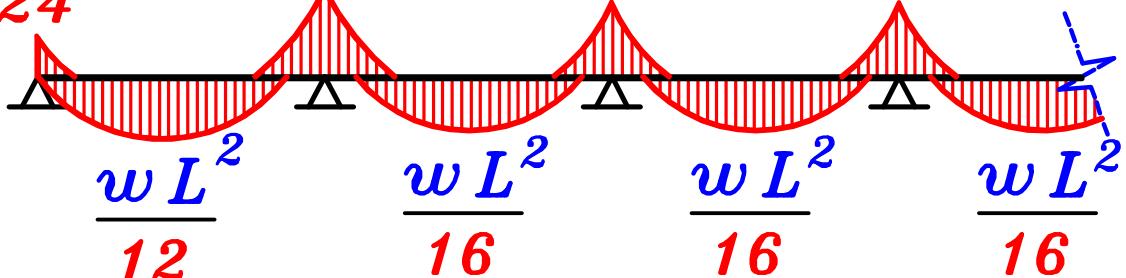
*Load For Moment.*

$$w \text{ kN/m}$$



$$\frac{wL^2}{24} \quad \frac{wL^2}{10} \quad \frac{wL^2}{12} \quad \frac{wL^2}{12}$$

*B.M.D.*

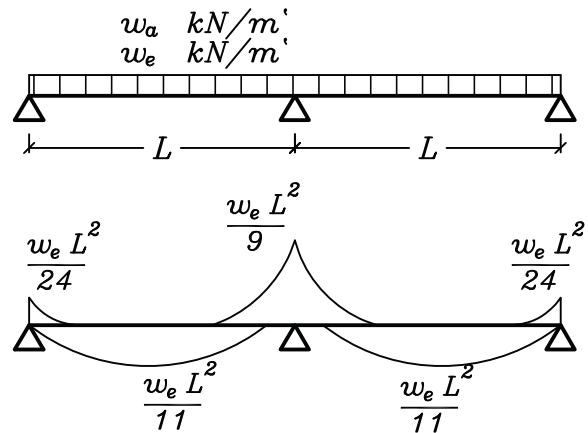


ملحوظه لا نعمل حالات تحميل للكمرات الـ *Continuos T.L.* و لكن نضع عليها.

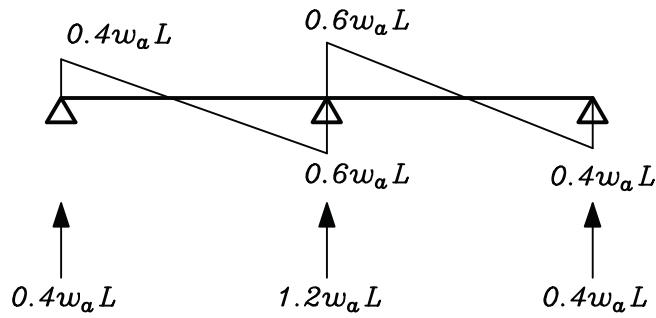
لان قيم الـ *max-max* محفوظه على أساس أن قيمة *w* هي *T.L.*

## Beams 2 Spans

B.M.D.

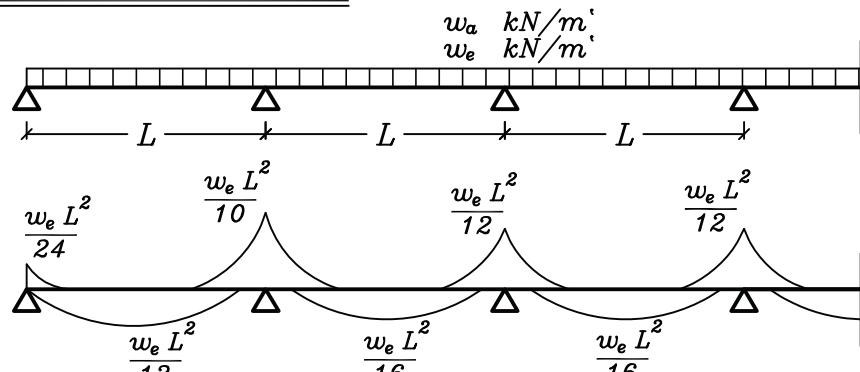


S.F.D.

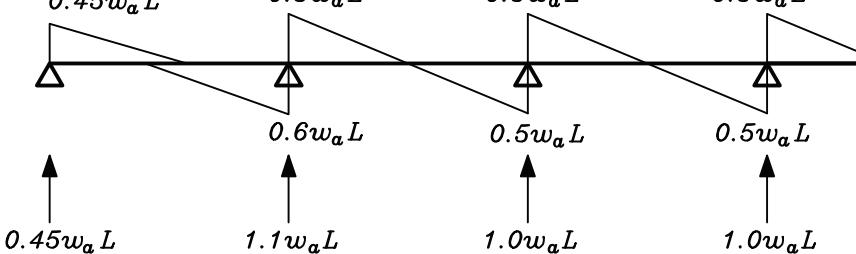


## Beams more than 2 Spans

B.M.D.

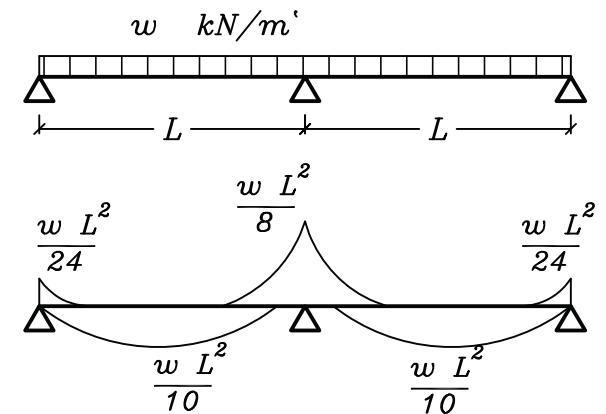


S.F.D.



## Slabs 2 Spans

B.M.D.



## Slabs more than 2 Spans

B.M.D.

